Flight test report

Manufacturer ADVANCE Thun AG

Address Seestrasse 14 3602 Thun Switzerland

Representive Kari Eisenhut Type of glider Alpha 4 28 not available Trimmer

PG 061.2007 Certification number Date of flight test 29/03/2007 Villeneuve Place of test



Classification B

Test Pilot Claude Thurnheer Harness sup air light

Total weight in flight 78 kg

Alain Zoller Advance - Progress L 106 kg

		Min weight	Max weight
. Inflation/Tal			
	Rising behaviour		A Smooth, easy and constant rising
Landina	Special take off technique required	No	A No
Landing	Special landing technique required	No	A No
Speed in st	raight flight	NO /	NO NO
ороса ст	Trim speed more than 30 km/h	Yes	A Yes
	Speed range using the controls larger than 10 km/h		A Yes
	Minimum speed	Less than 25 km/h	A Less than 25 km/h
Control mo			
	Max. weight in flight up to 80 kg		
	Symmetric control pressure/travel	Increasing, Greater than 55 cm	A not available
	Max. weight in flight 80 kg to 100 kg	and any Salata	O and avellable
	Symmetric control pressure/travel	not available	0 not available
	Max. weight in flight greater than 100 kg Symmetric control pressure/travel	not available	0 Increasing, Greater than 65 cm
Pitch stabili	ity exiting accelerated flight	not available	o increasing, Greater than 65 cm
non olubin	Dive forward angle on exit	Dive forward less than 30°	A Dive forward less than 30°
	Collapse occurs		A No
itch stabili	ity operating controls during accelerated flight		
	Collapse occurs	No	A No
Roll stabilit	y and damping		
	Oscillations	Reducing	A Reducing
tability in	gentle spirals		
ahavi	Tendency to return to straight flight	Spontaneous exit	A Spontaneous exit
senaviour i	n a steeply banked turn	40 / 4 4.4 /	A Mara than 4.4 m/s
Symmotric	Sink rate after two turns	12 m/s to 14 m/s	A More than 14 m/s
Symmetric	Entry	Rocking back less than 45°	A Rocking back less than 45°
	Recovery		A Spontaneous in less than 3 s
	Dive forward angle on exit		A Dive foward 0°to 30°, Keeping course
	Cascade occurs		A No
	With accelerator		
	Entry	Rocking back less than 45°	A Rocking back less than 45°
	Recovery		A Spontaneous in less than 3 s
	Dive forward angle on exit		A Dive foward 0°to 30°, Keeping course
	Cascade occurs		A No
Exiting de	ep stall (parachutal stall)		
	Deep stall achieved		A Yes
	Recovery		A Spontaneous in less than 3 s
	Dive forward angle on exit		A Dive forward 0°to 30°
	Change of course	0 0	A Changing course less than 45°
I Park and all	Cascade occurs	No	A No
High angle	e of attack recovery	Countains in less than 2 a	A Constant and in land than 2 a
	Recovery Cascade occurs		A Spontaneous in less than 3 s A No
Recovery	rom a developed full stall	INO	A INO
Recovery	Dive forward angle on exit	Dive forward 0°to 30°	A Dive forward 0°to 30°
	Collapse		A No collapse
	Cascade occurs (other than collapse)		A No
	Rocking back		A Less than 45°
	Line tension		A Most line tight
Asymmetr	ic collapse		
,	With 50% collapse-Maximum dive forward or roll angle		
	Change of course until re-inflation	Less than 90°, Dive or roll angle 0° to 15°	Less than 90°, Dive or roll angle 0° to 15°
	Re-inflation behaviour		A Spontaneous re-inflation
	Total change of course		A Less than 360°
	Collapse on the opposite side occurs	No	A No
	Twist occurs	No	A No
	Cascade occurs	No	A No
	With 75% collapse-Maximum dive forward or roll angle		
	Change of course until re-inflation		Less than 90°, Dive or roll angle 0° to 15°
	Re-inflation behaviour	•	A Spontaneous re-inflation
	Total change of course		A Less than 360°
	Collapse on the opposite side occurs		A No
	Twist occurs		A No
	Cascade occurs		A No
	With 50% collapse and accelerator-Maximum dive forward or		
	Change of course until re-inflation	Less than 90°, Dive or roll angle 0° to 15°	A Less than 90°, Dive or roll angle 0° to 15°
		Less than 90°, Dive or roll angle 0° to 15° Spontaneous re-inflation	A Less than 90°, Dive or roll angle 0° to 15° A Spontaneous re-inflation A Less than 360°
	Change of course until re-inflation	Less than 90°, Dive or roll angle 0° to 15°	

Twist occurs No A No Cascade occurs No A No	
	A
	Α
With 75% collapse and accelerator-Maximum dive forward or roll angle	
Change of course until re-inflation Less than 90°, Dive or roll angle 15° to 45° A Less than 90°, Dive or roll	
Re-inflation behaviour Spontaneous re-inflation A Spontaneous re-inflation	A
Total change of course Less than 360° A Less than 360°	A
Collapse on the opposite side occurs No A No	A
Twist occurs No A No	A
Cascade occurs No A No	A
15. Directional control with a maintained asymmetric collapse	_
Able to keep course Yes A Yes	A
180° turn away from the collapsed side possible in 10 s Yes A Yes	A
Amount of control range between turn and stall or spin More than 50 % of the symmetric control travel A More than 50 % of the symmetric	metric control travel A
16. Trim speed spin tendency	
Spin occurs No A No	A
17. Low speed spin tendency	
Spin occurs No A No	A
18. Recovery from a developed spin	
Spin rotation angle after release Stops spinning in less than 90° A Stops spinning in less than 90° Stops spinning in less spin spin spin spin spin spin spin sp	
Cascade occurs No A No	Α
19. B-line stall	
Change of course before release Change of course less than 45° A Change of course less than 45°	
Behaviour before release Remains stable with straight span A Remains stable with straight span A Remains stable with straight span	
Recovery Spontaneous in less than 3 s A Spontaneous in less tha	s A
Dive forward 0° to 30° A Dive forward 0° to 30° A Dive forward 0° to 30°	Α
Cascade occurs No A No	Α
20. Big ears	
Entry procedure Dedicated controls A Dedicated controls	Α
Behaviour during big ears Stable flight A Stable flight	Α
Recovery Spontaneous in less than 3 s A Spontaneous in less tha	Ss A
Dive forward 0° to 30° A Dive forward 0° to 30° A Dive forward 0° to 30°	Α
21. Big ears in accelerated flight	
Entry procedure Dedicated controls A Dedicated controls	Α
Behaviour during big ears Stable flight A Stable flight	Α
Recovery Spontaneous in less than 3 s A Spontaneous in less than	s A
Dive forward angle on exit Dive forward 0° to 30° A Dive forward 0° to 30°	Α
Behaviour immediately after releasing the accelerator while Stable flight A Stable flight	Α
22. Behaviour exiting a steep spiral	
Tendency to return to straight flight Spontaneous exit A Spontaneous exit	Α
Turn angle to recover normal flight Less than 720°, spontaneous recovery A Less than 720°, spontaneous recovery	
Sink rate when evaluating spiral stability [m/s] 19 m/s 16 m/s	
23. Alternative means of directional control	
180° turn achievable in 20 s Yes A Yes	Α
Stall or spin occurs No A No	Α
24. Any other flight procedure and/or configuration described in the user's manual	
Procedure works as described not available 0 not available	0
Procedure suitable for novice pilots not available 0 not available	0
Cascade occurs not available 0 not available	0
Comments of test pilot	
Comments no no	



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